

AMENDMENT TO SPECIFICATION

Replace paragraph 32 with the following, with markings to show current changes relative to the immediately prior version.

[0032] Figures 3-5 display one embodiment of the present invention. Figure 3 shows a stator vane cluster 101, which forms one segment of a stator vane stage of a gas turbine engine. The vane cluster 101 has one or more airfoils 103 extending between one or more platforms 105 (for clarity, Figure 3 only shows the inner platform). The platforms 105 define the radial boundaries of the annular core gas path through the engine 10. The airfoils 103 have a suction side 107 and a pressure side 109. The clusters 101 are similar to clusters 29. Namely, the clusters 101 have a fillet F between the airfoil 103 and the platforms 105 as a result of the machining process. In addition, the airfoils 103 have stagnation lines S, gage points G<sub>s</sub> on the suction sides 107 and gage points G<sub>p</sub> on the pressure sides 109.